



#### Torque Moment about longitudinal axis Corresponding deformation produced is twist or torsion.



Torque can be resisted in two different ways

- 1. Pure Torsion (St. Venant Torsion)
- 2. Warping Torsion

#### Pure Torsion

In this case the various cross-sections along the length of the member rotate relative to each other causing twist of the member. **Any particular cross section twists as a whole** Typical example is the torque applied on a circular rod. **Warping Torsion** The whole cross-sections do not rotate as a whole







Torsion Formula for Circular Section (Pure Torsion)

- 1. Plane section remains plane.
- 2. Radial lines remain straight.
- 3. Moment is applied along longitudinal axis.
- 4. Material remains elastic.

Torsion Formula for Circular Section (contd...)

 $\theta$  = total rotation of any section w.r.t. the reference point.

- $\phi$  = change of angle per unit length
- $\phi = \theta/L$  for linear increase  $\phi = d\theta/dz$  in general  $\rho$  = radial distance up to any point where stresses are to be calculated.
- $\tau$  = shear stress at any point
- $\gamma$  = shear strain at any point

Helix, deformed position of line AB after twist





Torsion Formula for Circular Section (contd...)

ρ

Shear stress due to pure torsion is always perpendicular to the redial distance at that point.

 $\gamma = \frac{B'C'}{dz} \longrightarrow \gamma = \frac{\rho d\theta}{dz}$  $\gamma = \rho \frac{d\theta}{dz} \longrightarrow \gamma = \rho \phi$ 





Torsion Formula for Circular Section (contd...)

 $dT = \rho \times \tau dA$  $= \rho \times (\gamma G) dA$   $\therefore \tau = \gamma G$  $= \rho \times (\rho \phi) \times \mathrm{Gd}A$  $= \rho^2 \times \phi \times \text{Gd}A$  $T = T_r = \int \rho^2 \times \phi \times \mathrm{Gd}A$  $T = \phi \times G \int \rho^2 dA$ 

- $T = \phi \times GJ$
- GJ = torsional rigidity
- $J = I_x + I_y$  For circular section
- Now  $\tau = \gamma G$

$$\tau = \rho \left(\frac{T}{GJ}\right) G$$

$$\tau = \frac{T \times \rho}{J}$$



Torsion Formula for Circular Section (contd...)



#### Pure Torsion For Non-Circular Section

#### (Experimental Method)

Soap Film Analogy

The volume between the bubble and the original plane (by the analogy of governing differential equation) is proportional to the total torque resistance (applied). Steeper the slope of tangent at any point greater will be the shear stress.

SFA is more useful for noncircular and irregular section for which formulas are difficult to derive.







Pure Torsion For Non-Circular Section Soap Film Analogy (contd...)



Soap Film



Pure Torsion For Non-Circular Section **By Timoshenko** 



#### **Plastic Torsion**

Whole the section will yield in torsion,  $\tau = \tau_v$ 

Plastic analysis assumes **uniform shear intensity** all around the surface and all around the cross section.



Plastic torsions can be envisioned in terms of SAND HEAP ANALOGY



#### Sand Heap Analogy

Put sand on a plate having a shape same as that of cross section (Circular, Rectangular, Irregular)





Sand Heap Analogy (contd...)

Volume under the sand heap is proportional to the torque.

$$\tau_{p_{\max}} = \frac{Tt}{\alpha_p b t^3}$$

$$\alpha_p = 0.33 \text{ for b/t} = 1.0$$
  
= 0.5 for b/t =  $\infty$ 



Torsion in Hollow Tubes

V = Resultant shear force at a face

 $\tau_1$  remains constant throughout the length

$$\begin{split} V_{AB} &= \tau_1 \times \mathbf{t}_1 \times dz \\ V_{CD} &= \tau_4 \times \mathbf{t}_2 \times dz \\ \sum F_z &= 0 \implies V_{AB} = V_{CD} \\ \tau_1 \times \mathbf{t}_1 &= \tau_4 \times \mathbf{t}_2 \end{split} \ \ \, \text{To maintain equilibrium} \end{split}$$

For equilibrium of infinitesimal element at corner B,  $\tau_1 = \tau_2$ 

Similarly, at corner C,  $\tau_3 = \tau_4$ 



Torsion in Hollow Tubes (contd...)



Shear stress is more in the portion where thickness is less but  $\tau \ge t$  remains constant

The product  $\tau \times t$  is referred to as the shear flow, *q* having units of N/mm. The shear flow remains constant around the perimeter of the tube.

This term comes from an analogy to water flowing in a loop of pipes having different diameters, where the total discharge remains the same.

Torsion in Hollow Tubes (contd...)



 $\tau \times t = q$  (Shear flow)

$$q_{\rm B} = q_{\rm C}$$

In general shear flow is same throughout the cross section.

Torsional shear force acting on *ds* length of wall =  $q \times ds$ 

Resisting moment of this force =  $r \times q \times ds$ 

Integrating this differential resisting torque around the perimeter gives the total resisting torque.

$$T = \int_{P} r \times q \times ds$$

Torsion in Hollow Tubes (contd...)

$$T = q \int_{P} r \times ds$$
$$T = 2q \int_{P} \frac{r \times ds}{2}$$
$$T = 2q \times A_{o}$$

 $A_o$  = Area enclosed by shear flow path

$$T = 2\tau \times t \times A_{o}$$
$$\tau = \frac{T}{2A_{o}t}$$

For hollow closed tube





#### Shear Center



"Shear center is defined as the point in the cross-sectional plane of a beam through which the transverse loads must pass so that the beam bends without twisting."

In other words, loads applied through the shear center will cause no torsional stresses to develop.



Shear Center (contd...)

$$T = \int_{0}^{n} (\tau \times t) \times r \times ds = 0$$





**Closed Thin Walled Section** 

Magnitude of Shear Flow for Transverse Loads Through Shear Center

 $q = \frac{VQ}{I}$  (1) Valid for sections having  $I_{xy} = 0$ 

"I" is about the axis of bending

$$q = \frac{V_{y}}{I_{x}I_{y} - I_{xy}^{2}} \left( I_{y} \int_{0}^{s} ytds - I_{xy} \int_{0}^{s} xtds \right)$$

If we put  $I_{xy} = 0$ , we will get (1)

Open Thin Walled Section



Shear Flow In Thin Walled Open Sections Due to Applied Shear Force

Rules For Plotting Shear Flow Diagram

- 1. The shear flow in the part of element parallel to the applied shear is always in a direction opposite to this applied shear.
- 2. Shear flow due to direct shear occurs in one direction through thin walls of open sections.
- 3. At junction of elements, incoming shear flow is equal to outgoing shear flow.





Rules For Plotting Shear Flow Diagram (contd...)

- 4. The value of shear flow is zero at free tips of the element and more shear flow is generated as more area is added.
- 5. Shear flow is assumed to be generated on one side of the neutral axis and consumed/absorbed on the other side.
- 6. Shear flow generated is proportional to the first moment of the area added.

- 7. Shear flow increases linearly for the elements perpendicular to the load and parabolically for the elements parallel to the load.
- 8. Shear flow is considered zero for elements which have insignificant contribution in corresponding "I" value.



Rules For Plotting Shear Flow Diagram (contd...)





General Rules For Locating Shear Center

- 1. Shear center always lie on axis of symmetry.
- 2. If two axes of symmetry exist for a section, S.C. will be at the intersection of these two axis.
- 3. If the centerlines of all the elements of a section intersect at a single point this is the shear center.
- 4. Shear center of "Z" section is at the centroid.





#### Procedure to Locate Shear Center

- 1. To find horizontal location ( $e_x$ ) apply vertical load (V) at  $e_x$  from reference point.
- 2. Plot shear flow diagram due to applied load.
- 3. Find the internal shear force in each element.
- 4. Apply  $\Sigma M = 0$  at convenient location and find  $e_x$
- 5. Similarly apply horizontal load at a vertical distance " $e_y$ " from reference point (say centroid) and repeat the above procedure to calculate " $e_y$ "
- 6. The distances " $e_x$ " and " $e_y$ " locate the shear center.

Example:

Locate the Shear Center for the given channel section.

 $d \longrightarrow \begin{bmatrix} t_{w} \\ t_{w} \\ t_{w} \end{bmatrix} \qquad h = d - t_{f}$  $b = b_{f} - \frac{t_{w}}{2}$ 

**Centerline Representation** 

#### Solution



By symmetry about z-axis, the shear center must lie at half the depth. Only horizontal location is to be found.



Point A

$$Q = (b \times t_f) \times \frac{h}{2}$$
$$q_A = \frac{V}{I_x} (b \times t_f) \times \frac{h}{2}$$

Point P

$$q_{P} = q_{A} + \frac{V}{I_{x}} \times \left(\frac{h}{2} \times t_{w}\right) \times \frac{h}{4}$$
$$q_{P} = \frac{V}{I_{x}} \times \left(bt_{f} \times \frac{h}{2} + \frac{h^{2}}{8} \times t_{w}\right)$$

 $\mathbf{i}$ 





Solution

 $\Sigma M_{P} = 0$  $V \times e_x - V_f \times \frac{h}{2} - V_f \times \frac{h}{2} = 0$  $V \times e_x = V_f \times h$  $e_x = \frac{h}{V} \left( \frac{V}{I_x} \times \frac{b^2 t_f h}{4} \right)$  $b^2 t_f h^2$  $e_x$ 

Positive means on the assumed left side.



Solution





Differential Equation for Torsion of I-Shaped Sections

u<sub>f</sub> = lateral deflection of one of the flanges

 $\theta$  = twist angle at the selected section

V<sub>f</sub> = Shear force in flange due to torsion. (internal force developed)

 $\boldsymbol{\theta}$  is smaller and is in radians, so

$$u_{f} \cong \theta \times \frac{h}{2}$$
 (1)


Differential Equation for Torsion of I-Shaped Section (contd...)

The lateral curvature relationship of one flange alone is:

$$\frac{d^{2}u_{f}}{dz^{2}} = -\frac{M_{f}}{EI_{f}}$$
 (2)

M<sub>f</sub> = Lateral Bending moment on one flange

I<sub>f</sub> = Moment of inertia of one flange about y-axis of beam

$$I_f = \frac{t_f b_f^3}{12}$$



Differential Equation for Torsion of I-Shaped Section (contd...)

$$V = \frac{dM}{dz} \Longrightarrow V_f = \frac{dM_f}{dz}$$
(3)

Differentiating (1)

$$\frac{d^3 u_f}{dz^3} = \frac{-V_f}{EI_f} \quad (4)$$

$$V_f = -EI_f \frac{d^3 u_f}{dz^3}$$

$$V_f = -EI_f \frac{(h/2)d^3\theta}{dz^3} \implies V_f = -EI_f \frac{h}{2} \frac{d^3\theta}{dz^3}$$
(5)



Differential Equation for Torsion of I-Shaped Section (contd...)

Torsion resistance due to warping



Warping Constant  $C_{w} = I_{f} \frac{h^{2}}{2}$   $C_{w} = \frac{t_{f} b_{f}^{3}}{12} \times \frac{h^{2}}{2}$  $C_w = \frac{I_y}{2} \times \frac{h^2}{2}$ 

(6)



Differential Equation for Torsion of I-Shaped Section (contd...)

Torsion resistance due to Pure torsion

$$M_s = GJ \times \frac{d\theta}{dz}$$

For Circular Section



For Non-Circular Section

Total Torque Applied  $M_z = M_s + M_w$  $M_z = GC \times \frac{d\theta}{dz} - EC_w \frac{d^3\theta}{dz^3}$ (8)

OR



Differential Equation for Torsion of I-Shaped Section (contd...)

Dividing by "- $EC_w$ "

where

 $\frac{d^{3}\theta}{dz^{3}} - \frac{GJ}{EC_{w}} \times \frac{d\theta}{dz} = -\frac{M_{z}}{EC_{w}} \qquad (9)$   $\frac{d^{3}\theta}{dz^{3}} - \lambda^{2} \frac{d\theta}{dz} = -\frac{M_{z}}{EC_{w}} \qquad (10)$ Non homogeneous differential equation  $\lambda^{2} = \frac{GC}{EC_{w}} \qquad \qquad \lambda = \sqrt{\frac{GC}{EC_{w}}} \qquad (11)$ 

 $\lambda^2$  = Ratio of pure torsion rigidity to warping torsion rigidity



Differential Equation for Torsion of I-Shaped Section (contd...)

**Total Solution** 

$$\theta = \theta_h + \theta_P$$

 $\theta$  = Total Solution  $\theta_{h}$  = Homogeneous Solution  $\theta_{P}$  = Particular Solution

Homogeneous Equation

$$\frac{d^3\theta}{dz^3} - \lambda^2 \frac{d\theta}{dz} = 0$$



Differential Equation for Torsion of I-Shaped Section (contd...)

**Trial Function** 

$$\theta_h = A e^{mz}$$

"A", "m" are constants. "z" is independent variable

$$\frac{d^3\theta}{dz^3} = Am^3 e^{mz}$$



$$Am^{3}e^{mz} - \lambda^{2} \times Ame^{mz} = 0$$
$$Ae^{mz} (m^{3} - \lambda^{2}m) = 0$$

For non-trivial solution  $A \neq 0$ 



Differential Equation for Torsion of I-Shaped Section (contd...)

$$m^{3} - \lambda^{2}m = 0$$
$$m(m^{2} - \lambda^{2}) = 0$$

Possible Solutions:  $m = 0, m = +\lambda, m = -\lambda$ 

Sum of all solutions is total homogeneous solution

$$\theta_h = A_1 e^{\lambda z} + A_2 e^{-\lambda z} + A_3 e^o$$
$$= A_1 e^{\lambda z} + A_2 e^{-\lambda z} + A_3$$

We know

 $\sinh(x) + \cosh(x) = e^x$  and  $\sinh(x) - \cosh(x) = e^{-x}$ 



Differential Equation for Torsion of I-Shaped Section (contd...)

$$\theta_{h} = A_{1} [\sinh(\lambda z) + \cosh(\lambda z)] + A_{2} [\cosh(\lambda z) - \sinh(\lambda z)] + A_{3}$$
$$\theta_{h} = \sinh(\lambda z) (A_{1} - A_{2}) + \cosh(\lambda z) (A_{1} + A_{2}) + A_{3}$$
$$\theta_{h} = A \sinh(\lambda z) + B \cosh(\lambda z) + C \qquad (13)$$
Homogeneous solution



Differential Equation for Torsion of I-Shaped Section (contd...)

#### Particular solution

Consider  $M_z$  to be constant or linearly varying along the length

 $M_z = f(z)$  [Constant or function of first degree].  $\theta_p$  may assumed to be a polynomial of degree up to 2, as twist due to pure torque is first integral of moment.

Let

$$\theta_P = f_1(z) - (14)$$
e.g.  $f_1(z) = Dz^2 + Ez + F$ 
Uniform torque

Polynomial of second order. One order higher that applied torque.



Differential Equation for Torsion of I-Shaped Section (contd...)

Try this particular integral in (10)

$$\frac{d^{3}f_{1}(z)}{dz^{3}} - \lambda^{2} \frac{df_{1}(z)}{dz} = -\frac{1}{EC_{w}} f(z) \qquad \text{As } M_{z} = f(z)$$
Polynomial of Ist order
$$\lambda^{2} \frac{df_{1}(z)}{dz} = \frac{1}{EC_{w}} f(z) \qquad (15)$$

Boundary conditions 1- Torsionally Simply Supported  $\theta = 0 \quad \frac{d^2\theta}{dz^2} = 0 \quad \frac{d\theta}{dz} \neq 0$ 



Flanges can bend laterally



Differential Equation for Torsion of I-Shaped Section (contd...)

This is equivalent to deflection and moment made equal to zero for simple support for bending. Change of twist  $d\theta / dz$  may have any value at the end.

Flange may displace at the end but web is held at its position.

2- Torsionally Fixed End

$$\theta = 0 \quad \frac{d^2\theta}{dz^2} \neq 0 \quad \frac{d\theta}{dz} = 0$$

The constant of integration will be evaluated for individual cases.







Differential Equation for Torsion of I-Shaped Section (contd...)

After getting the value of constants and full solution for  $\theta$ , the stresses may be evaluated as follows:

Pure Torsional Shear Stress

where

$$v_{s} = \frac{T}{C}$$
$$T = GC \frac{d\theta}{dz}$$
$$v_{s} = Gt \frac{d\theta}{dz}$$

Tr



(16)

Differential Equation for Torsion of I-Shaped Section (contd...)

#### Warping Shear Stress

No stress in the web





Differential Equation for Torsion of I-Shaped Section (contd...)

#### Normal Warping Stress

(in the flanges)









#### The design and allowable torsion strengths are below:

Design torsional strength in LRFD=  $\phi_t T_n$ Allowable torsional strength in ASD=  $T_n / \Omega_t$ Resistance factor for torsion in LRFD=  $\phi_t = 0.9$ Safety factor for torsion in ASD=  $\Omega_t = 1.67$ 



The nominal torsional strength  $(T_n)$  according to the limit states of torsional yielding and torsional buckling is:

$$T_n = F_n C$$

The following nomenclature may be used in the further discussion:

= 
$$2(B - t)(H - t) - 4.5(4 - \pi)t^3$$
 for  
rectangular HSS

$$C = \frac{\pi (D-t)^2 t}{2}$$
 for round HSS



- *B* = overall width of rectangular HSS
- *H* = overall height of HSS
- h = clear distance between the flanges
  less the inside corner radius on each
  side
- *D* = outside diameter of round HSS
- *L* = length of the member

#### F<sub>n</sub> For Round HSS

$$F_{\rm n} = F_{\rm cr} = \text{larger of} \quad \frac{1.23E}{\sqrt{\frac{L}{D}\left(\frac{D}{t}\right)^{5/4}}} \quad \text{and} \quad \frac{0.60E}{\left(\frac{D}{t}\right)^{3/2}}$$

but the value should not exceed  $0.6F_{y}$ 

#### F<sub>n</sub> For Rectangular HSS

i) For 
$$\frac{h}{t} \leq 2.45 \sqrt{\frac{E}{F_y}}$$
  $F_n = F_{cr} = 0.6F_y$ 



ii) For 
$$2.45\sqrt{\frac{E}{F_y}} < \frac{h}{t} \le 3.07\sqrt{\frac{E}{F_y}}$$
  
 $F_n = F_{cr} = 0.6F_y \left(2.45\sqrt{\frac{E}{F_y}}\right)\frac{h}{t}$   
iii) For  $3.07\sqrt{\frac{E}{F_y}} < \frac{h}{t} \le 260$   
 $F_n = F_{cr} = \frac{0.458\pi^2 E}{(h/t)^2}$ 



#### **F**<sub>n</sub> For Other Sections

a) For the limit state of yielding under normal stress:

$$F_n = F_y$$

b) For the limit state of shear yielding under shear stress:

$$F_{\rm n} = 0.6F_{\rm y}$$

c) For the limit state of buckling

$$F_n = F_{cr}$$

where  $F_{\rm cr}$  for buckling is to be determined by detailed analysis.



#### Example:







Solution: (contd...)

$$f(z) = \frac{T}{2}$$

$$T = P \times e = 90 \times 50$$

$$= 4500 kN - mm$$
•  $\theta_P = C_1 + C_2 z = f_1(z)$ 
• One order ahead
$$\frac{d^3 f_1(z)}{dz^3} - \lambda^2 \frac{df_1(z)}{dz} = -\frac{1}{EC_w} f(z)$$
where
$$\lambda^2 = \frac{GJ}{EC_w}$$

$$0 - \lambda^2 (C_2) = -\frac{1}{EC_w} \frac{T}{2}$$
$$C_2 = \frac{T}{2} \frac{1}{EC_w} \times \frac{EC_w}{GJ} = \frac{T}{2GC}$$

#### So, the particular solution is:

$$\theta_P = C_1 + \frac{T}{2GC} \times z$$

Solution: (contd...)

The total solution is

$$\theta = A \sinh(\lambda z) + B \cosh(\lambda z) + C + C_1 + \frac{T}{2GC} \times z$$

$$\theta = A \sinh(\lambda z) + B \cosh(\lambda z) + \frac{T}{2GC} \times z + C$$

**Boundary Conditions** 

$$z = 0, \quad \theta = 0$$
 (II)  
 $z = 0, \quad \frac{d^2\theta}{dz^2} = 0$  (II)  $z = \frac{L}{2}, \quad \frac{d\theta}{dz} = 0$  (III)



Solution: (contd...)

To apply the boundary condition first we have to take I<sup>st</sup> and 2<sup>nd</sup> derivatives

$$\frac{d\theta}{dz} = A\lambda \cosh(\lambda z) + B\lambda \times \sinh(\lambda z) + \frac{T}{2GC}$$
$$\frac{d^{2}\theta}{dz^{2}} = A\lambda^{2} \sinh(\lambda z) + B\lambda^{2} \times \cosh(\lambda z)$$
$$\frac{d^{3}\theta}{dz^{3}} = A\lambda^{3} \cosh(\lambda z) + B\lambda^{3} \times \sinh(\lambda z)$$
$$(I) \Rightarrow 0 = 0 + B + 0 + C \implies B + C = 0$$
$$(II) \Rightarrow 0 = 0 + B\lambda^{2} \times 1 \implies B = 0$$
$$\implies C = 0$$



Solution: (contd...)

(III) 
$$\Rightarrow 0 = A\lambda \cosh\left(\lambda \frac{L}{2}\right) + 0 + \frac{T}{2GC}$$



$$\theta = \frac{-T}{2GC\lambda} \left( \frac{1}{\cosh(\lambda L/2)} \right) \times Sinh(\lambda z) + \frac{T}{2GC} z$$



Solution: (contd...)

$$\theta = \frac{T}{2GC\lambda} \left( \lambda z - \frac{\sinh(\lambda z)}{\cosh(\lambda L/2)} \right)$$
$$\frac{d\theta}{dz} = \frac{T}{2GC\lambda} \left( \lambda - \frac{\lambda \cosh(\lambda z)}{\cosh(\lambda L/2)} \right)$$
$$\frac{d\theta}{dz} = \frac{T}{2GC} \left( 1 - \frac{\cosh(\lambda z)}{\cosh(\lambda L/2)} \right)$$
$$\frac{d^2\theta}{dz^2} = \frac{T\lambda}{2GC} \left( \frac{-\sinh(\lambda z)}{\cosh(\lambda L/2)} \right) \qquad \frac{d^3\theta}{dz^3} = \frac{T\lambda^2}{2GC} \left( -\frac{\cosh(\lambda z)}{\cosh(\lambda L/2)} \right)$$



Solution: (contd...)

W 460 x 106

 $S_x = 2080 \times 10^3 \text{ mm}^3$   $I_x = 48,700 \times 10^4 \text{ mm}^4$   $C = J = 145 \times 10^4 \text{ mm}^4$   $C_w = 12,62,119 \times 10^6 \text{ mm}^6$   $1/\lambda = 1501 \text{ mm}$ L = 7500 mm

$$h = d - t_f = 448.4 mm$$
  
 $t_f = 20.6 mm$   
 $t_w = 12.6 mm$   
 $b_f = 194 mm$   
 $d = 469 mm$ 



#### Solution: (contd...)

| Z    | λz    | Sinh(λz) | Cosh(λz) |
|------|-------|----------|----------|
| 0    | 0     | 0        | 1.000    |
| 0.1L | 0.5   | 0.521    | 1.128    |
| 0.2L | 0.999 | 1.174    | 1.542    |
| 0.3L | 1.499 | 2.127    | 2.350    |
| 0.4L | 1.999 | 3.623    | 3.759    |
| 0.5L | 2.498 | 6.038    | 6.120    |

 $G = \frac{E}{2(1+\nu)} = \frac{2,00,000}{2(1+0.3)} = 76,923 \text{MPa}$  $GC = 76,923 \times 145 \times 10^4 = 1115 \times 10^8 N - mm^2$ 

Solution: (contd...)

Pure Torsional Shear Stress

$$v_{s} = Gt \frac{d\theta}{dz}$$

$$v_{s} = \frac{Tt}{2J} \left( 1 - \frac{\cosh(\lambda z)}{\cosh(\lambda L/2)} \right)$$

$$v_{s} = \frac{4500 \times 1000t}{2 \times 145 \times 10^{4}} \left( 1 - \frac{\cosh(\lambda z)}{\cosh(2.49)} \right)$$

$$v_{s} = 1.552t \left( 1 - \frac{\cosh(\lambda z)}{6.120} \right)$$



Solution: (contd...)

Maximum pure torsional shear stress is at the ends

$$(v_{s})_{\max_{z=0,L}} = 1.55t \left( 1 - \frac{\cosh(\lambda \times 0)}{6.120} \right)$$

$$(v_{s})_{\max_{z=0,L}} = 1.297t$$

$$(v_{s})_{\max_{for \text{ flange}}} = 1.297t_{f} = 1.297 \times 20.6 = 26.72MPa$$

$$(v_s)_{\max_{\text{for web}}} = 1.297 t_w = 1.297 \times 12.6 = 16.34 MPa$$



Solution: (contd...)

#### Warping Shear Stress

In flanges

$$(v_w)_{\max} = \frac{Eb_f^2 h}{16} \frac{d^3 \theta}{dz^3}$$
  
=  $\frac{Eb_f^2 h}{16} \times \frac{T\lambda^2}{2GC} \left(-\frac{\cosh(\lambda z)}{\cosh(\lambda L/2)}\right)$   
=  $\frac{T}{2C_w} \frac{b_f^2 h}{16} \left(-\frac{\cosh(\lambda z)}{6.12}\right)$   $\lambda^2 = \frac{GC}{EC_w}$   
=  $-0.307 \cosh(\lambda z)$ 



Solution: (contd...)

Along the length maximum value will occur at z=L/2

$$(v_w)_{\max}$$
 at midspan  $_{z=L/2} = -0.307 \cosh\left(\lambda \times \frac{L}{2}\right)$ 

$$= -1.88MPa$$

$$(v_w)$$
 at ends  $_{z=0} = -0.31MPa$ 



Solution: (contd...)

Normal Warping Stress

$$(f_{bw})_{\max} = \frac{Eb_f h}{4} \frac{d^2 \theta}{dz^2}$$
$$(f_{bw})_{\max} = \frac{Eb_f h}{4} \times \frac{T\lambda}{2GC} \left(-\frac{\sinh(\lambda z)}{6.12}\right)$$
$$(f_{bw})_{\max} = -9.56 \sinh(\lambda z)$$

As flanges are simply supported at ends, the maximum stress will be at mid-span

$$(f_{bw})_{\max} = -9.56 \sinh\left(\lambda \times \frac{L}{2}\right) = -57.69MPa$$



Solution: (contd...)

Maximum Normal Stress due to Ordinary Flexure

$$f_{b} = \frac{M}{S_{x}} = \frac{PL/4}{S_{x}}$$
$$= \frac{90,000 \times 7500/4}{2080 \times 10^{3}}$$
$$f_{b} = 81.13MPa$$


Solution: (contd...)

Shear Stress due to Ordinary Bending

$$v = \frac{VQ}{Ib}$$

At the N.A.:

$$v = \frac{45000 \left(194 \times 20.6 \times \frac{448.4}{2} + 12.6 \times \frac{427.8}{2} \times \frac{427.8}{4}\right)}{48,700 \times 10^4 \times 12.6}$$

$$v = 8.68MPa$$





Solution: (contd...)

### At face of Web:

With in flange at edge of web

$$v = \frac{VQ}{Ib}$$

$$v = \frac{45000 \left(\frac{194 - 12.6}{2} \times 20.6 \times \frac{448.4}{2}\right)}{48,700 \times 10^4 \times 12.6} = 1.88 MPa$$

$$v = 1.88MPa$$





#### Summary of Stresses



Results: Beam is safe in flexure, torsion and shear at all the sections





Analogy Between Warping Torsion and Lateral Bending



 $P_{\rm H} \mathbf{x} h = T$ 

 $P_{\rm H} = T/h$ 

### Analogy For Torsion (contd...)



• Because the differential equation solution is time consuming, and really suited only for analysis, design of a beam to include torsion is most conveniently done by making the analogy between torsion and ordinary bending

- It is assumed that all the torque is resisted by warping torsion which is not the actual situation (solution will be approximate).
- $\beta$  factor is used to reach near to actual solution.
- β factors are problem specific values, depending on end conditions.
- Tables have been proposed for β factor to cover different situations.
- $\beta$  factor tables are available on Page # 476 & 477, (Salmon & Johnson)

$$\beta_1$$
  $\beta_2$ 

Analogy For Torsion (contd...)





### Example:

Select a W section for a beam to carry 9kN/m dead load including the self weight, and a live load of 24 kN/m. The load is applied at an eccentricity of 175mm from center of web. The simply supported span is 8.0 m. Assume that ends of beam are simply supported for torsion.

Solution:

$$v_u = 1.2D + 1.6L$$
  
= 1.2 × 9 + 1.6 × 24  
= 49.2 kN/m

### Solution: (contd...)





Solution: (contd...)

$$M_{ux} = \frac{49.2 \times 8.0^2}{8} = 393.60 \ kN - m$$
$$m_u = 49.2 \times \frac{175}{1000} = 8.61 \ kN - m / m$$
$$d_{\min} = \frac{L}{22} = \frac{8000}{22} = 364mm$$

Let  $h \cong 364mm$ 

Assume  $\lambda L = 3.0$  Initial assumption



Solution: (contd...)

$$M_{f} = \beta \frac{w_{H}L^{2}}{8}$$
$$M_{f} = \beta \frac{m_{u}}{h} \times \frac{L^{2}}{8} \qquad \qquad w_{H} = \frac{m_{u}}{h}$$

2

z = 0.5L, a = 0.5

From table 8.6.8, P # 477

$$\lambda L = 3 \Longrightarrow \beta = 0.51$$
  
 $M_f = 0.51 \times \frac{8.61}{0.364} \times \frac{8.0^2}{8} = 96.51 \text{ kN} - \text{m}$ 



Solution: (contd...)

$$(S_x)_{req} = \frac{M_{ux}}{\phi_b F_y} + \frac{2M_f (S_x / S_y)}{\phi_b F_y}$$
Approximate value
$$(S_x)_{req} = \frac{393.60 \times 10^6}{0.9 \times 250} + \frac{2 \times 96.51 \times 10^6 (2.75)}{0.9 \times 250}$$

$$(S_x)_{req} = 4109 \times 10^3 \, mm^3$$

### Solution: (contd...)

Where high torsional strength is required, W360 sections are preferable because these usually give less stresses due to torsional warping. Check conditions of compact section

#### **Trial Section**

### W 360 x 237



### Solution: (contd...)

1

Assuming that  $L_b \leq L_{p'}$  no problem of LTB

$$\lambda L = \frac{1}{1735} \times 8000 = 4.61$$

$$\beta = 0.27 + \frac{0.37 - 0.27}{1} \times (5.0 - 4.61) = 0.309$$

$$h = d - t_f = 380 - 30.2 = 349.8mm$$

$$M_f = \beta \frac{m_u}{h} \times \frac{L^2}{8}$$

$$= 0.309 \times 8.61 \times \frac{8.0^2}{8} \times \frac{1}{0.3498} = 60.85kN - m$$

| λL | β    |
|----|------|
| 4  | 0.37 |
| 5  | 0.27 |



### Solution: (contd...)

Normal Bending Stress At Mid-span



= 171.64 MPa

 $<\phi_b F_y = 225 MPa$  **O.K.** 



Solution: (contd...)

### Shear Stress

- Warping torsion.....Critical at center
- Vertical bending.....Critical at ends
- Pure torsion.....Critical at ends.

Warping Shear Stress At Mid-Span:

$$V_{w} = \frac{V_{f}Q_{f}}{I_{f}t_{f}}$$
$$V_{f} = \beta \frac{m_{u}}{h} = 0.309 \times \frac{8.61 \times 1000}{0.3498} = 7606 \text{ N}$$



Solution: (contd...)

$$Q_{f} = \frac{b_{f}}{2} \times t_{f} \times \frac{b_{f}}{4} = 589 \times 10^{3} mm^{3}$$

$$I_{f} = \frac{I_{y}}{2} = 15550 \times 10^{4} mm^{4}$$

$$v_{w} = \frac{7606 \times 589 \times 10^{3}}{15550 \times 10^{4} \times 18.9} = 1.524 MPa \quad <135 MPa$$
O.K.

Total shear stress = 1.524 MPa, because there is no applied shear at the center and there is no simple torsion **O.K.** 



Solution: (contd...)

Web Shear Stress (end section):

$$v = \frac{(wL/2) \times Q}{I_x \times t_w}$$
At N.A.  

$$Q = \left(b_f \times t_f \times \frac{h}{2}\right) + t_w \left(\frac{h - t_f}{2}\right) \times \left(\frac{h - t_f}{4}\right) = 2328 \times 10^3 mm^3$$

$$v = \frac{(49.2 \times 8.0/2) \times 1000 \times 2328 \times 10^3}{79100 \times 10^4 \times 18.9}$$

$$= 30.65 MPa$$



Solution: (contd...)

Pure Torsion (end section):

 $v_{s} = \frac{T \times t_{w}}{C}$   $T = \frac{m_{u}L}{2} = 34.44 \text{ kN} - \text{mm}$   $V_{s} = \frac{34.44 \times 10^{6} \times 18.9}{824 \times 10^{4}} = 79.00 MPa$ 

Total Shear stress at end section = 30.65 + 79.00 = 109.65MPa<135MPa O.K.



Flange Shear Stress (end section):

$$v = \frac{(wL/2) \times Q}{I_x \times t_w}$$

At Junction of Web and Flange

$$Q = \frac{b_f}{2} \times t_f \times \frac{b_f}{4} = 1043 \times 10^3 \, mm^3$$
$$v = \frac{196.8 \times 1000 \times 1043 \times 10^3}{79100 \times 10^4 \times 30.2}$$
$$= 8.59 MPa$$



Pure Torsion (end section):

$$v_{s} = \frac{T \times t_{w}}{C}$$

$$T = \frac{m_{u}L}{2} = 34.44 \text{ kN} - \text{mm}$$

$$V_{s} = \frac{34.44 \times 10^{6} \times 30.2}{824 \times 10^{4}} = 113.62MPa$$



Including small warping contribution in the same formula

Total shear stress at end section =  $v + v_s + v_s$ 

| Table. Values of $\lambda$ and C |             |                                       |
|----------------------------------|-------------|---------------------------------------|
| Designation                      | $1/\lambda$ | $C = J  (\times 10^4 \text{ mm}^{4)}$ |
| W360 × 216                       | 1869        | 633                                   |
| × 237                            | 1735        | 824                                   |
| × 262                            | 1600        | 1100                                  |
| × 287                            | 1483        | 1450                                  |
| × 314                            | 1389        | 1860                                  |
| × 347                            | 1288        | 2480                                  |
| × 382                            | 1196        | 3290                                  |
| × 421                            | 1118        | 4330                                  |
| × 463                            | 1046        | 5660                                  |
| × 509                            | 980         | 7410                                  |
| × 551                            | 932         | 9240                                  |
| × 592                            | 892         | 11400                                 |
| × 634                            | 853         | 13800                                 |





### Concluded